

But the second term in Eq. (35) must not be small. Multiplying both sides of Eq. (34) by  $X^T M$  gives

$$X^T M \bar{F}^{-1} = -(1/\varepsilon) X^T \quad (36)$$

Inserting Eq. (36) into Eq. (35) yields

$$\bar{F}^{-1} = G^+ + X X^T M \bar{F}^{-1} \quad (37)$$

i.e.,

$$G^+ = (I - X X^T M) \bar{F}^{-1} \quad (38)$$

Similarly, to avoid the calculation of  $\bar{F}^{-1}$  in Eq. (38), an interim solution  $\tilde{v}_j$  is first computed in Eq. (39):

$$\bar{F} \tilde{v}_j = q_j \quad (39)$$

i.e.,

$$\tilde{v}_j = \bar{F}^{-1} q_j \quad (40)$$

Embedding Eq. (37) into Eq. (40) gives

$$v_j = (I - X X^T M) \tilde{v}_j \quad (41)$$

Both Eqs. (39) and (41) are two principal formulas, as shown in Ref. 9. It is concluded that the generalized inverse technique described in this Note is also one of the DP methods as shown in Ref. 9. This is known by the fact that  $MUDU^T M$  (or  $MXDX^T M$ ) in Eq. (11) is considered as the perturbation term of DP method in Ref. 9.

## VI. Numerical Results

The particular solution  $v_j$  of a numerical example shown in Ref. 8 is computed by using the GIT method in this paper and the scalar constant  $c_j$  in the general solution  $y_j' = v_j + Y c_j$  of eigenvector derivatives is given by the procedure stated in Refs. 7 and 8, in which  $Y = [y_1, y_2, \dots, y_m]$ . The results (general solution  $y_j'$ ) are the same as those of Refs. 8 and 9. In addition, to address the effect of different values of  $d_j$ , we select  $d_j = d = 0.000001, 1.0$ , and  $100,000$  ( $j = 1, 2, \dots, m$ ), in which  $d_j = d$  means that all  $d_j$  ( $j = 1, 2, \dots, m$ ) are taken as a uniform value  $d$ . It is found that for very different  $d$ , the obtained solution  $y_j'$  is the same and that the particular solution  $v_j$  equals to the general solution  $y_j'$ . The former tells us that this method is numerically stable. The latter tells us that the particular solution  $v_j$  obtained by this method possibly happens to be just the general solution  $y_j'$ . Thus, it is very meaningful to propose a simple criterion by which we can know whether the  $v_j$  given by this method is  $y_j'$ . Finally, the results with  $u_j = y_j$  and  $u_j = x_j$  are the same.

## VII. Conclusions

Based on technical analysis and numerical results, a new approach for determining the eigenvector derivatives with repeated roots has been successfully developed. This method can be applied to the calculation of eigenvector derivatives with and without repeated roots and avoids the process exerting one or  $m$  constraints to the governing equation of the eigenvector derivative. This method requires only the knowledge of the eigenvalue and its associated eigenvector under consideration. This method is numerically stable and requires no numerical convergence checks when incorporated into the existing computer program.

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# Distributed Modeling and Actuator Location for Piezoelectric Control Systems

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## Introduction

THE piezoelectric materials have been used extensively in disturbance sensing and control. Since the piezoelectric materials are distributed (not discrete) in nature, an accurate sensing and control of structures can be achieved by piezoelectric sensors and actuators.

The finite element method is a powerful technique widely used in many modern engineering design and analysis problems. Since the governing equations of piezoelectric media are complex in general, the finite element modeling suitable for the thin piezoelectric media becomes important.<sup>1</sup> The problem of sensor and actuator placement exists with piezoelectric sensors and actuators if they are discretely placed on various members. Hence, the issue of the piezoelectric sensor and actuator problem has been addressed<sup>2</sup> to see the effect of the placement on the control efficiency.

In this work, two-dimensional thin rectangular elements with pseudointernal degrees of freedom (DOF) are used to develop the finite element model of piezoelectric materials. The internal DOF are for the better representation of bending moments generated by the feedback piezoelectric voltage and are condensed to the physical nodal DOF using the Guyan reduction technique. The control efficiency effect of the location of the piezoelectric actuator pair on the beam is studied.

## Finite Element Modeling of Piezoelectric Media

The quasistatic piezoelectric equations are given as<sup>3</sup>

$$\begin{aligned} T &= cS - eE \\ D &= e^T S + \varepsilon E \end{aligned} \quad (1)$$

where  $T$ ,  $S$ ,  $E$ , and  $D$  are the vectors of stress, strain, electric field, and charge per unit area and  $c$ ,  $e$ , and  $\varepsilon$  are the matrices of

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constitutive coefficients. The following relation is also given for the electric field in the piezoelectric media:

$$\mathbf{E} = -\nabla\phi \quad (2)$$

where  $\phi$  is the electric potential. A functional  $\Pi$  is defined as<sup>4</sup>

$$\begin{aligned} \Pi = & \int_V G \, dV - \int_V \mathbf{u}^T \mathbf{P}_b \, dV \\ & - \int_{S_1} \mathbf{u}^T \mathbf{P}_s \, dS - \mathbf{u}^T \mathbf{P}_c + \int_{S_2} \phi \sigma \, dS \end{aligned} \quad (3)$$

where  $G$  is the thermopiezoelectric potential,  $\mathbf{P}_b$  and  $\mathbf{P}_s$  are the respective vectors of body and surface forces applied to volume  $V$  and surface  $S_1$ ,  $\mathbf{P}_c$  is the vector of concentrated forces,  $\mathbf{u}$  is the displacement vector, and  $\sigma$  is the surface charge on the surface  $S_2$ . Hamilton's principle is used to obtain the finite element formulation of the piezoelectric media. The principle is expressed as

$$\delta \int_{t_1}^{t_2} (T - \Pi) \, dt = 0 \quad (4)$$

where the kinetic energy  $T$  is given by

$$T = \frac{1}{2} \int_V \rho \dot{\mathbf{u}}^T \dot{\mathbf{u}} \, dV \quad (5)$$

where  $\dot{\mathbf{u}}$  is the velocity vector. Equation (4) is now written as

$$\delta \int_{t_1}^{t_2} \Pi \, dt = \delta \int_{t_1}^{t_2} T \, dt = - \int_{t_1}^{t_2} \int_V \rho \delta \mathbf{u}^T \ddot{\mathbf{u}} \, dV \, dt \quad (6)$$

upon using integration by parts with proper boundary conditions. For the thermodynamic potential  $G$ ,

$$\delta G = \delta \mathbf{S}^T \mathbf{T} - \delta \mathbf{E}^T \mathbf{D} \quad (7)$$

For the finite element formulation, the rectangular elements ( $2a \times 2b$ ) with pseudointernal DOF are used to better represent the bending moments caused by the piezoelectric actuator. The term  $N_i$  ( $i = 1, \dots, 4$ ) is the  $i$ th shape function used for the rectangular elements, and the other two shape functions for the pseudointernal DOF are given as

$$N_5 = 1 - (x^2/a^2), \quad N_6 = 1 - (y^2/b^2) \quad (8)$$

which vanish at the element boundaries where  $x = \pm a$  and  $y = \pm b$ . Applying the finite element method to Eq. (4) yields the following piezoelectric equations (after assemblage):

$$\begin{bmatrix} M_{uu} & 0 \\ 0 & 0 \end{bmatrix} \begin{Bmatrix} \ddot{\mathbf{u}} \\ \ddot{\phi} \end{Bmatrix} + \begin{bmatrix} K_{uu} & K_{u\phi} \\ K_{\phi u} & -K_{\phi\phi} \end{bmatrix} \begin{Bmatrix} \mathbf{u} \\ \phi \end{Bmatrix} = \begin{Bmatrix} \mathbf{F} \\ \mathbf{G} \end{Bmatrix} \quad (9)$$

where the Guyan static model reduction technique is used to condense the pseudointernal DOF to the physical nodal DOF. In Eq. (9),  $\mathbf{u}$ ,  $\phi$ ,  $\mathbf{F}$ , and  $\mathbf{G}$  are the global nodal displacement, electric potential, force, and applied charge vectors, respectively. The element matrices and vectors in Eq. (9) are found as

$$\begin{aligned} [M_{uu}]_{\text{cl}} &= \int_{V_{\text{cl}}} \rho N_u^T N_u \, dV, & [K_{u\phi}]_{\text{cl}} &= \int_{V_{\text{cl}}} B_u^T e B_\phi \, dV \\ [K_{uu}]_{\text{cl}} &= \left[ \int_{V_{\text{cl}}} B_u^T c B_u \, dV \right] - \left[ \int_{V_{\text{cl}}} B_u^T c Y \, dV \right] \\ &\quad \times \left[ \int_V Y^T c Y \, dV \right]^{-1} \left[ \int_{V_{\text{cl}}} B_u^T c Y \, dV \right]^T \\ [K_{\phi\phi}]_{\text{cl}} &= \int_{V_{\text{cl}}} B_\phi^T \epsilon B_\phi \, dV, & \mathbf{G}_{\text{cl}} &= \int_{(S_2)_{\text{cl}}} N_\phi^T \sigma \, dS \\ \mathbf{F}_{\text{cl}} &= \int_{V_{\text{cl}}} N_u^T \mathbf{P}_b \, dV + \int_{(S_1)_{\text{cl}}} N_u^T \mathbf{P}_s \, dS + N_u^T \mathbf{P}_c \end{aligned} \quad (10)$$

where  $N_u$  and  $N_\phi$  are the shape function matrices for displacement and electric fields and  $B_u$  and  $B_\phi$  are defined for each element as

$$B_u = [L_u \quad N_u], \quad B_\phi = [\nabla N_\phi] \quad (11)$$

where  $L_u$  is the differential operator that relates strain to displacement in two dimensions. In Eq. (10),  $X$  and  $Y$  matrices are given for each element as

$$X = \begin{bmatrix} 0 & 0 \\ N_5 & N_6 \end{bmatrix}, \quad Y = [L_u X] \quad (12)$$

### Effect of Piezoelectric Actuator Location in Active Control

A cantilever beam having a piezoelectric actuator pair at the top and bottom surfaces is considered as a system to investigate the effect of distributed piezoelectric actuator location in active vibration control. The sensed signals are fed to the piezoelectric pair according to the control law implemented. The linear quadratic regulator (LQR) scheme is employed as the control law to determine the control gains necessary for adjusting sensory signals. The equations of motion of the system are converted to the standard state-space form as

$$\dot{\mathbf{z}} = \mathbf{A}\mathbf{z} + \mathbf{B}\mathbf{u}_G \quad (13)$$

Equation (9) can be written in the form of Eq. (13) with

$$\mathbf{z} = \begin{Bmatrix} \mathbf{u} \\ \dot{\mathbf{u}} \end{Bmatrix}, \quad \mathbf{u}_G = \mathbf{V} = (1/c_p)\mathbf{G} \quad (14)$$

$$\mathbf{A} = \begin{bmatrix} 0 & \mathbf{I} \\ -\mathbf{M}_{uu}^{-1} \mathbf{K}_n & 0 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} 0 \\ \mathbf{M}_{uu}^{-1} \mathbf{D} \end{bmatrix}$$

where

$$\mathbf{K}_n = \mathbf{K}_{uu} + \mathbf{K}_{u\phi} \mathbf{K}_{\phi\phi}^{-1} \mathbf{K}_{\phi u}, \quad \mathbf{D} = c_p \mathbf{K}_{u\phi} \mathbf{K}_{\phi\phi}^{-1} \quad (15)$$

and  $c_p$  is the capacitance. According to the LQR scheme, the control input vector  $\mathbf{u}_G$  is chosen of the form

$$\mathbf{u}_G = -\mathbf{K}_G \mathbf{z} \quad (16)$$

so that the performance index

$$J = \int_0^\infty (\mathbf{z}^T \mathbf{Q} \mathbf{z} + \mathbf{u}_G^T \mathbf{R} \mathbf{u}_G) \, dt \quad (17)$$

with respective positive semidefinite and positive definite weighting matrices of  $\mathbf{Q}$  and  $\mathbf{R}$  is minimized. The scheme yields the gain matrix  $\mathbf{K}_G$  to be

$$\mathbf{K}_G = \mathbf{R}^{-1} \mathbf{B}^T \mathbf{P} \quad (18)$$

where  $\mathbf{P}$  is given by the algebraic Riccati equation

$$\mathbf{P} \mathbf{A} + \mathbf{A}^T \mathbf{P} - \mathbf{P} \mathbf{B} \mathbf{R}^{-1} \mathbf{B}^T \mathbf{P} + \mathbf{Q} = 0 \quad (19)$$

In this research study, the location of the piezoelectric actuator pair is varied along the cantilever beam to see the impact of piezoelectric actuator location on the vibration control efficiency of the closed-loop system. The closed-loop system here refers to the cantilever beam controlled by the piezoelectric actuator. For the control efficiency, the absolute sum of the tip deflections of the closed-loop system subjected to some disturbance within a time interval is taken as objective. The magnitude of this objective is computed for different actuator locations. When the magnitude of this objective is less at a certain actuator location, say at  $d_1$ , compared with another location, say at  $d_2$ , it is decided that the actuator pair at location  $d_1$  is more effective than that at  $d_2$ .

## Numerical Results

The cantilever beam with piezoelectric actuator pair shown in Fig. 1 is used to illustrate the procedure just outlined. The piezoelectric material is taken as polyvinylidene fluoride (PVDF), and the beam is made of aluminum. The material properties of PVDF are  $c_{11} = 3.8 \times 10^9$  Pa,  $e_{31} = 0.046$  C/m<sup>2</sup>,  $\epsilon_{33} = 1.062 \times 10^{-10}$  F/m,  $c_p = 3.8 \times 10^{-6}$  F, and  $\rho = 1800$  kg/m<sup>3</sup>; those of aluminum are  $c_{11} = 7.3 \times 10^{10}$  Pa,  $\rho = 2750$  kg/m<sup>3</sup>, and  $\nu = 0.33$ . The heights of the beam and piezoelectric actuators are taken as  $h_b = 6$  and  $0.6$  mm. The depth of both beam and piezoelectric actuators is  $w = 6$  mm, and the length of the beam is  $L = 60$  mm. The length of both piezoelectric actuators is assumed as  $L/2$ .

The closed-loop structure controlled according to the LQR scheme is subjected to a unit step force at the tip in the  $y$  direction, and the simulation results are given in Figs. 2 and 3. Figure 2 shows the tip deflection (tip displacement in the  $y$  direction) of

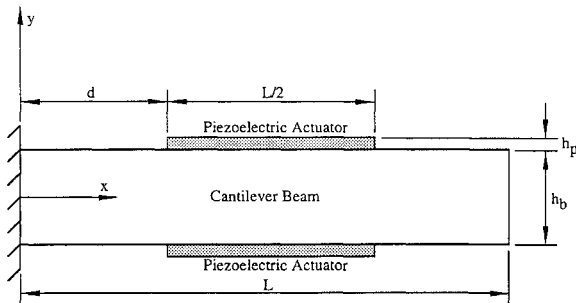


Fig. 1 Cantilever beam with piezoelectric actuator pair (not to scale).

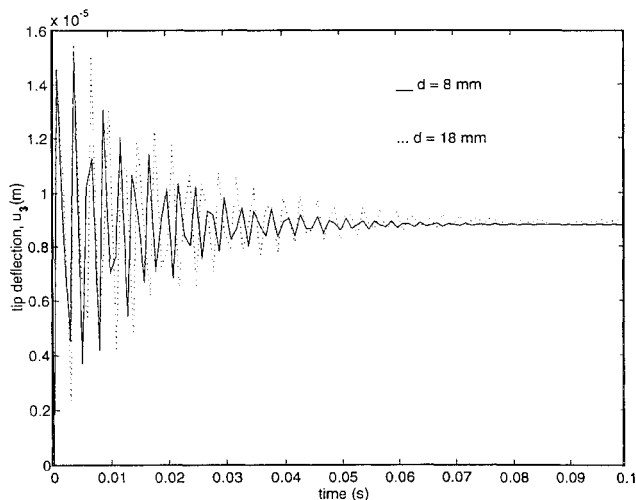


Fig. 2 Vibrational tip deflection of closed-loop system.

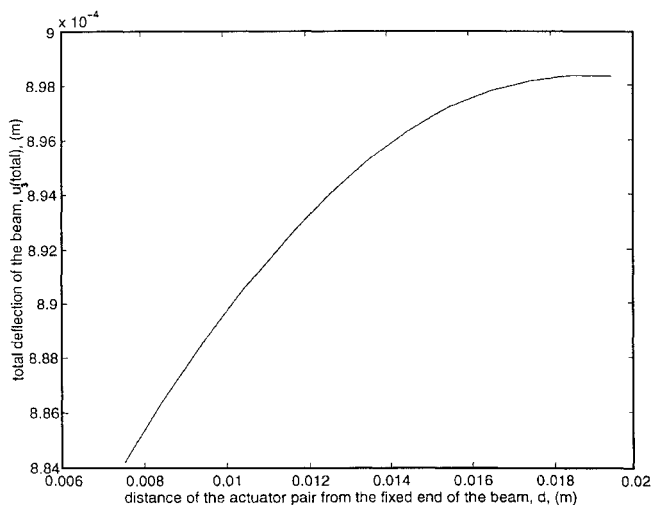


Fig. 3 Absolute sum of vibrational tip deflections of closed-loop system from 0 to 0.1 s.

the beam for two different actuator locations, whereas Fig. 3 shows the absolute sum of the tip deflections of the beam in the given time interval (the sum is found by adding the absolute magnitudes at every 0.001 s). The  $Q$  and  $R$  matrices for the LQR method are taken as  $0.001 \times I$  and  $I$ , where  $I$  is the identity matrix with proper dimensions. Other  $Q$  and  $R$  matrices can be used equally well. It is evident from the results, especially in Fig. 3, that the piezoelectric actuator pair closer to the fixed end is more effective in attenuating structural vibrations than the pair further away from the fixed end.

## Conclusion

Observing the magnitude of structural vibrations, the piezoelectric actuator pair closer to the fixed end of the cantilever beam is more efficient than the pair further away from the fixed end. This result is important in piezoelectric actuator placement for structures that can be modeled as cantilever beams. The procedure outlined in this Note can be generalized to more complicated systems.

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# Practical Complete Modal Space and Its Applications

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## I. Introduction

THE modal expansion method has been used in many important subjects, such as dynamic response, substructural modal synthesis, calculation of eigenvector derivatives, model correction, and reduction of dynamic models. In using the modal expansion technique, there exists numerical error as a result of truncation of modes, and the precision of the results sometimes is poor. To reduce the error of truncated modes, the development of a practical complete modal space (PCMS) method is necessary and strongly recommended.

The PCMS method was first applied to the correction of structural dynamic models.<sup>1–4</sup> Later, it was successfully used in the calculation of the eigenvector derivatives with repeated frequencies,<sup>5</sup> in the development of an accurate modal synthesis,<sup>6–9</sup> in the reduction of dynamic models,<sup>10,11</sup> and in the medium substitution problems for

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